

**AAE 340 – Dynamics and Vibrations**  
**Problem Set 2**  
**Due: 1/23/13**

**Problem 1:** A particle P departs along the curve originating from point O.

(a) Note that unit vectors  $\hat{e}$  are inertial;  $\hat{u}$  move and are cylindrical. Write an expression for the vector  $\vec{r}^{OP}$ . (Note that it can be written in terms of both  $r$  and  $\theta$ .) Express  $\vec{r}^{OP}$  in terms of both  $\hat{e}$  and  $\hat{u}$ .

(b) Consider the velocity  ${}^{\infty}\vec{v}^{OP}$ . Derive an expression for this velocity using  $\hat{e}$  as the working frame; write the final expression for  ${}^{\infty}\vec{v}^{OP}$  in terms of both  $\hat{e}$  and  $\hat{u}$ .

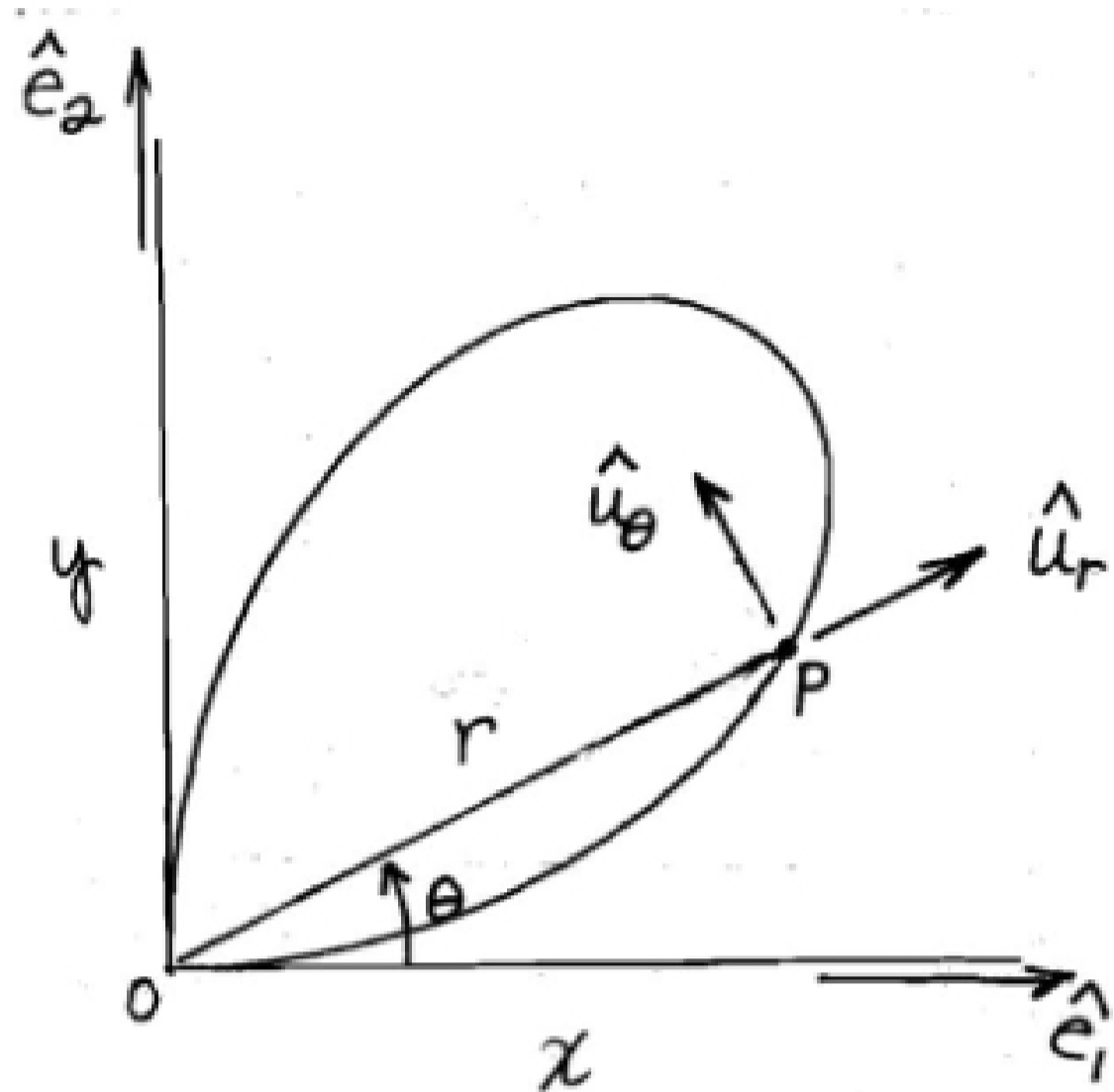
(c) Derive an expression for  ${}^{\infty}\vec{v}^{OP}$  using  $\hat{u}$  as the working frame; write the final expression for  ${}^{\infty}\vec{v}^{OP}$  in terms of both  $\hat{e}$  and  $\hat{u}$ .

Are the expressions for  ${}^{\infty}\vec{v}^{OP}$  in (b) and (c) the same? Should they be the same?

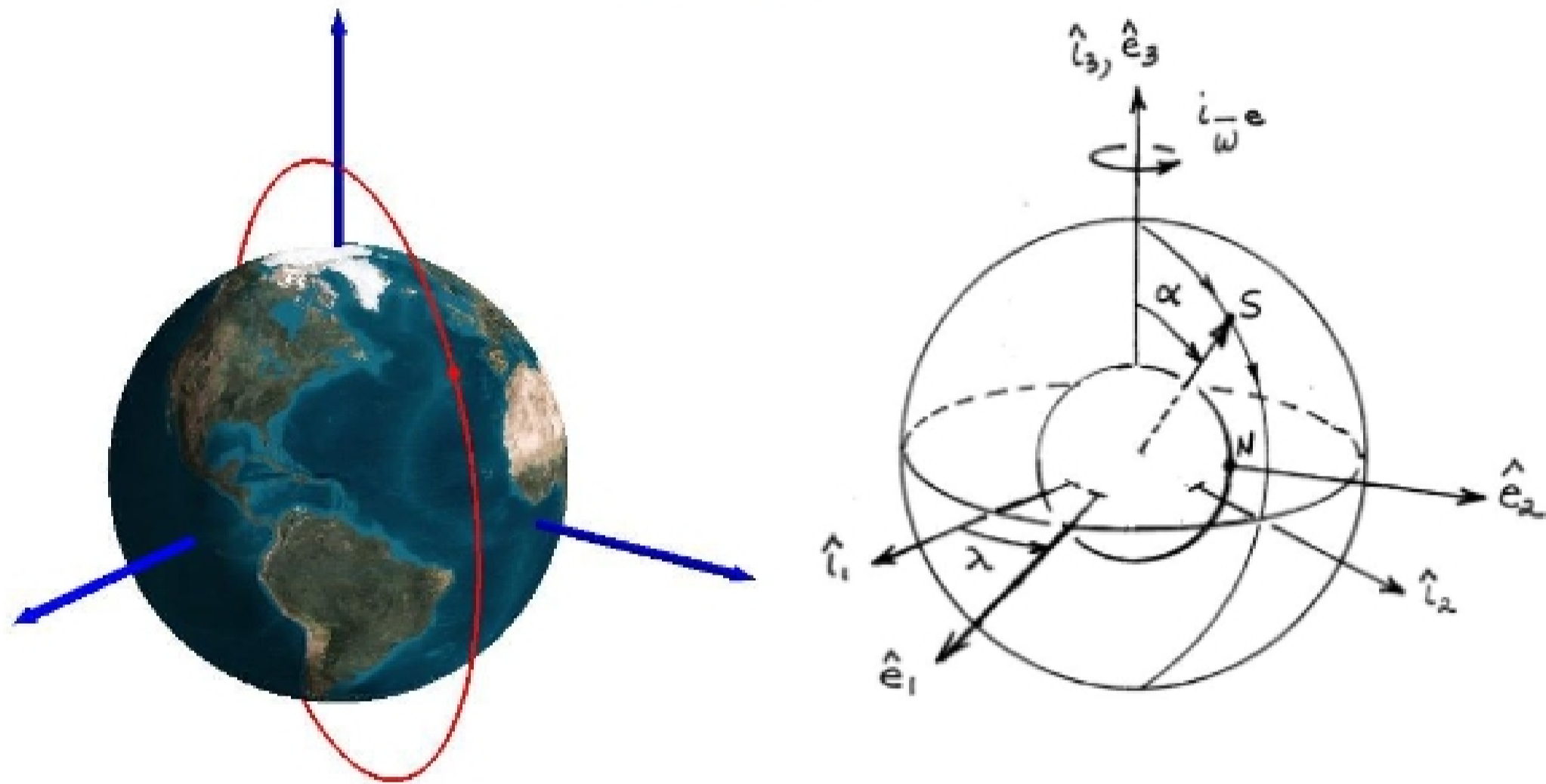
Is  ${}^{\infty}\vec{v}^{OP}$  a generic velocity? How do you know?

(d) The curve in the figure is defined by the equation  $r = C \sin(2\theta)$  where  $C$  is a constant and  $\dot{\theta}$  is a constant value. At  $t = 0$ ,  $\theta = 0^\circ$ . When time equals 6 seconds, and if  $C = 4$  ft and  $\dot{\theta} = 10$  deg/s, evaluate the expressions for  ${}^{\infty}\vec{v}^{OP}$  (the expression in  $\hat{e}$  as well as  $\hat{u}$ ).

Compute the magnitude of  ${}^{\infty}\vec{v}^{OP}$ . Is the magnitude the same for both expressions, i.e.,  ${}^{\infty}\vec{v}^{OP}$  expressed in  $\hat{e}$  as well as  $\hat{u}$ ? Should it be the same?



**Problem 2:** A satellite  $S$  is moving in a circular, polar orbit about the Earth (red path on left). The constant orbital radius (from the center of the Earth) is 7000 km; the satellite velocity relative to an inertial observer  $(\hat{i}_1, \hat{i}_2, \hat{i}_3)$  is given in terms of a constant magnitude 7.456 km/s. Note that any orbit plane is always fixed in the inertial frame, in this case the  $\hat{i}_2 - \hat{i}_3$  plane. Let  $\hat{e}_1, \hat{e}_2, \hat{e}_3$  (blue axes on left) be fixed in the Earth and, thus, this set of unit vectors rotates with the Earth; the Earth rotates on its own axis once every 23.934 hours about the axis  $\hat{i}_3 = \hat{e}_3$ . Thus, the angle between  $\hat{i}_1$  and  $\hat{e}_1$  is  $\lambda$  and  ${}^i\omega^e = \frac{2\pi \text{ rad}}{23.934 \text{ hrs}}$ . Note that the radius of the Earth is 6378 km.



- Sketch the orbit of the satellite as it appears in the  $\hat{i}_2 - \hat{i}_3$  plane. Define a set of unit vectors  $\hat{s}_j$  that move with the satellite. Write expressions for the unit vector relationships between  $\hat{i}$  and  $\hat{e}$  as well as  $\hat{e}$  and  $\hat{s}$ . (Note that  $\hat{s}$  is not truly a spherical set of unit vectors!)
- Assume that point  $O$  is located at the center of the Earth. Write the position vector  $\vec{r}^{OS}$  and express it in terms of each of the three sets of unit vectors  $\hat{e}, \hat{i}, \hat{s}$ .
- Use the BKE and derive kinematical expressions for  ${}^i\vec{v}^{OS}, {}^i\vec{A}^{OS}$ . Express both in terms of  $\hat{s}_j$ . Using the given information, determine the numerical values for  $\dot{\alpha}, \ddot{\alpha}$  and evaluate  ${}^i\vec{v}^{OS}, {}^i\vec{A}^{OS}$ . (Are these generic?)
- Return to the full sketch above. Recall that unit vectors  $\hat{e}_1, \hat{e}_2, \hat{e}_3$  are fixed in the Earth. Use the BKE and derive general expressions for velocity and acceleration relative to an

observer fixed on the Earth, that is,  ${}^e\bar{v}^{OS}$  and  ${}^e\bar{A}^{OS}$ . Express these quantities in terms of  $\hat{e}$ ;  $\hat{s}$ .

- (c) Express  ${}^i\bar{v}^S$ ,  ${}^i\bar{A}^S$  in terms of unit vectors fixed in the Earth. Are these the same quantities as  ${}^e\bar{v}^S$ ,  ${}^e\bar{A}^S$ ? Should they be equal?
- (d) The point N is identified as an observer fixed on the surface of the Earth at the location indicated — perhaps a tracking station. Write the position vector  $\bar{r}^{NS}$ .

Derive an expression for  ${}^e\bar{v}^{NS} = \frac{d\bar{r}^{NS}}{dt}$ . How does this expression compare to the velocities  ${}^i\bar{v}^{OS}$ ,  ${}^e\bar{v}^{OS}$ ? Are they the same? Does that make sense?

Evaluate this velocity  ${}^e\bar{v}^{NS} = \frac{d\bar{r}^{NS}}{dt}$ . What is the speed of the satellite relative to the tracking station at the instant when  $\alpha = 60^\circ$  and  $\lambda = 45^\circ$ ?